We consider plane and axisymmetric hypersonic flows of an ideal gas behind a compression shock satisfying the conditions $\cos (\mathbf{n}, \mathbf{i}) \leqslant \tau, \cos (\mathbf{n}, \mathbf{j}) \sim 1$ everywhere except for a small vicinity of the apex. Here the small parameter $\tau \ll 1 ; \mathbf{n}$ is the normal to the surface of the front; $i$ is. the unit vector of the $x_{1}$ axis; $j$ is the unit vector of the $x_{2}$ axis; $L x_{i}$ ( $i=1,2$ ) is the rectangular Cartesian coordinate system; $L$ is the characteristic length. The impinging stream is assumed to be homogeneous. Assuming that the condition $K=M_{\infty} \tau \geqslant 1$ is satisfied for a shock having close to a power-law form close to the apex, we consider the behavior of the solution outside the vicinity of the apex as $\tau \rightarrow 0$ on the basis of the method of joined asymptotic expansions. In doing this we take the expansion of the hypersonic theory of small perturbations [1], which describes unsteady one-dimensional flows in the zeroth approximation, as the outer expansion in the region adjacent to the front.

A great many results in the theory of the flow over thin bodies by a hypersonic stream are based on an analogy with unsteady flows in a space with a number of dimensions smaller by one. However, the validity of such an analogy in the case when an entropy layer develops in the stream has not yet been fully proven [2]. A detailed analysis of the known results on this question is carried out in [3, 4]. An explicit connection between the two-dimensional problem under consideration and the unsteady one-dimensional problem is established in the present report. Particular cases of the results obtained are compared with the known results.

The system of gasdynamic equations in the variables "pressure-two stream functions" [5] has the following form in the two-dimensional case:

$$
\begin{gather*}
\frac{\partial u_{i}}{\partial p}=(-1)^{i} x_{2}^{v} \frac{\partial x_{i+1}}{\partial \psi} ; \frac{\partial x_{1}}{\partial p} u_{2}=\frac{\partial x_{2}}{\partial p} u_{1}  \tag{1}\\
\frac{u_{1}^{2}-u_{9}^{9}}{2}+\frac{x}{x-1} \frac{p}{\rho}=\frac{1}{2}+\frac{1}{x-1} \mathrm{M}_{\infty}^{-2} ; \frac{\partial}{\partial p}\left(\frac{p}{\rho^{x}}\right)=0 .
\end{gather*}
$$

Here and below an index $i>2$ is read as $i-2 ; v=0$ and 1 for the plane and axisymmetric cases; $\rho_{\infty}{ }^{2}{ }_{\infty} p$ is the pressure; $\rho_{\infty} \rho$ is the density; $u_{\infty} u_{i}$ are the components of the velocity vector along the $X_{i}$ axes, respectively; $\rho_{\infty} \mu_{\infty} L^{1}+v_{\psi}$ is the stream function; the $X_{1}$ axis is directed along the velocity vector $u_{\infty}$ of the undisturbed stream; and $k$ is the ratio of specific heat capacities of the gas. Henceforth we will assume that the condition $1<k<2$ is satisfied. The index $\infty$ pertains to the conditions in the undisturbed stream.

The boundary conditions at the shock wave have the form

$$
\begin{gathered}
p=\frac{2}{\alpha+1}(\mathbf{i} \cdot \mathbf{n})^{2}+\frac{1-\alpha}{\alpha(\alpha+\mathbf{1})} \mathrm{M}_{\infty}^{-2} ; \frac{1}{\rho}=\frac{\kappa-1}{x+1}+\frac{2}{x+1} \mathrm{M}_{\infty}^{-2}(\mathbf{i} \cdot \mathbf{n})^{-2} ; \\
(\mathbf{i}-\mathbf{v})(\mathbf{i} \cdot \mathbf{n})=\left(p-\frac{1}{\alpha} \mathrm{M}_{\infty}^{-2}\right) \mathbf{n},
\end{gathered}
$$

where $v$ is the velocity vector of the disturbed motion at the shock. Being confined for simplicity to a surface of a front which is symmetrical relative to the $x_{1}$ axis, we assign the analytical expression for the shape of the front in the form

$$
x_{2}=\tau 2^{v / 2} \zeta^{1 /(1+v)}, x_{1}=g(\zeta) \text { for } x_{2} \geqslant 0
$$

Novosibirsk. Translated from Zhurnal Prikladnoi Mekhaniki i Tekhnicheskoi Fiziki, No. 6, pp. 38-48, November-December, 1976. Original article submitted October 15, 1975.

[^0]where the function $g(\zeta)$ provides for a smooth solution in the disturbed region and has the following properties: 1) $\lim g(\zeta) / \zeta^{\alpha}=$ const as $\zeta \rightarrow 0 ; 1 /(1+v)<\alpha \leqslant(3+v) / 2(1+v)$; 2) $\mathrm{dx}_{2} / \mathrm{dx}_{1} \leqslant 0(\tau)$ everywhere except for a small vicinity of the apex. Here the limitation $\alpha>1 /(1+v)$ provides for the appearance of an entropy layer, while $\alpha \leqslant(3+v) / 2(1+v)$ is a limitation emerging from the mean of construction of the solution outside the entropy layer. The first property assumes that the form of the front is close to a power law at the apex. Outside the vicinity of the apex in the region where $p \leqslant 0\left(\tau^{2}\right)$ the solution is constructed by the method of joined asymptotic expansions [6].

In the inner region adjacent to the streamline $\zeta=0$ the inner expansion is constructed as follows. We consider the entropy layer, i.e., the gas layer passing through the section of the shock wave with a slope $1>\beta \geqslant 0, t^{\beta} \leqslant \cos (\mathbf{n}, \mathbf{i}) \leqslant 1$. With allowance for the law of conservation of entropy per particle, as well as the order of magnitude of the pressure, we obtain in the entropy layer $\left.0\left(\tau^{2 / x}\right) \leqslant \rho \leqslant 0\left(\tau^{(2}-2 \beta\right) / x\right)$. Henceforth we set $\rho \sim 0(\tau \gamma)$ in the entropy layer, where $2 / k \geqslant \gamma \geqslant(2-2 \beta) / k$. At the boundary of the region under consideration $u_{2} \leqslant 0(\tau)$. Assuming that the condition $u_{2} \leqslant 0(\tau)$ is satisfied across the region under consideration, we obtain from the Bernoulli integral

$$
\begin{equation*}
u_{1} \sim 1+0\left(\tau^{2}-\gamma\right) . \tag{2}
\end{equation*}
$$

Thus,

$$
\begin{equation*}
u_{1}=f\left(1+O\left(\tau^{2}\right)\right), f=\{1-[2 \kappa /(\varkappa-1)] p / \rho\}^{1 / 2} . \tag{3}
\end{equation*}
$$

And for the derivative of the Bernoulli integral with allowance for (2) we get

$$
\begin{equation*}
\partial u_{1} / \partial p=-(1 / \rho f)(1 \div 0(\tau v)) . \tag{4}
\end{equation*}
$$

Across the disturbed region $\gamma$ varies from $2 / \mathrm{k}$ in the vicinity of the surface of the body to 0 in the vicinity of the surface of the front. A concept of the region of applicability of (4) can be obtained by considering the isobars $p=$ const. For example, everywhere in the region under consideration where $\left|\partial \mathrm{p} / \partial \mathrm{x}_{1} / \partial \mathrm{p} / \partial \mathrm{x}_{2}\right| \geqslant 0(1)$, the parameter $\gamma \geqslant 1$.

The fulfillment of Eq. (4) in the entropy layer allows one to considerably simplify system (1). Substituting (3) and (4) into (1), we obtain

$$
\begin{align*}
& \left(\frac{x_{2}}{\tau}\right)^{1+v}=2^{v} \int_{0}^{\xi} \frac{1}{\rho f} d \zeta(1+0(\tau v))+x_{20}^{1}+v(\xi, \tau) ; \\
& \frac{\partial}{\partial \xi}\left[\frac{\partial}{\partial \xi}\left(\frac{x_{2}}{\tau}\right)\left(\frac{\partial x_{1}}{\partial \xi}\right)^{-1} f\right](1+0(\tau v))=\left(\frac{x_{2}}{\tau}\right)^{v} \frac{\partial x_{1}}{\partial \xi} ;  \tag{5}\\
& \frac{u_{2}}{\tau} \frac{\partial x_{1}}{\partial \xi}=f \frac{\partial}{\partial \xi}\left(\frac{x_{2}}{\tau}\right)\left(1+0\left(\tau^{2}\right)\right), \xi / \rho^{\kappa}=F(\tau, \xi),
\end{align*}
$$

where $p=\tau^{2} \xi ; \psi=\tau^{1+\gamma} \zeta ; x_{20}$ and $F$ are arbitrary functions. In the coordinate system used the pressure enters into the integrand as a parameter. Therefore, the integration of system (1), if the entropy distribution along the streamlines is known, comes down to the integration of the one quasilinear parabolic equation with respect to $x_{1}$ in system (5) in the entropy layer.

Representing $x_{1}$ for small $\zeta$ in the form of asymptotic expansions with respect to $\zeta$, we obtain from the second equation of system (5)

$$
\begin{equation*}
x_{1}=x_{10}(\xi, \tau)+x_{11}(\xi, \tau) \zeta+\varphi(\xi, \tau, \zeta), \tag{6}
\end{equation*}
$$

where the function $\mathrm{x}_{1}$ o is arbitrary;

$$
x_{11}=\left\{\begin{array}{l}
x_{20}^{-v} \frac{d}{d \xi}\left[\frac{d}{d \xi}\left(x_{20}\right) \frac{d x_{10}^{-1}}{d \xi}\right], \quad \text { if } \quad x_{20} \neq 0, \\
\nu \xi^{\frac{1}{2 x}} \frac{d}{d \xi}\left[\frac{d}{d \xi}\left(\xi^{-\frac{1}{2 \alpha}}\right) \frac{d x_{11}^{-1}}{d \xi}\right], \quad \text { if } \quad x_{20}=0 .
\end{array}\right.
$$

The function $\varphi$ does not exceed the following quantities in order of magnitude:

$$
\varphi=\left\{\begin{array}{l}
0\left(\tau^{2-\frac{2}{x}} \zeta+\tau^{\gamma} \zeta+\int_{0}^{\zeta} \int_{0}^{\rho} \frac{1}{\rho} d \zeta d \zeta\right), \quad \text { if } \quad x_{20} \neq 0 \\
0\left(\int_{0}^{\tau} \int_{0} \frac{1}{\rho} d \zeta d \zeta\right) \text { at } v=0 ; \quad 0\left(\tau^{2-\frac{2}{x}} \zeta+\tau_{\zeta}^{\gamma}\right) \\
\text { at } v=1, \text { if } \quad x_{20}=0
\end{array}\right.
$$

The approximate solution (3)-(6) constructed depends on the three arbitrary functions $\mathrm{X}_{20}(\xi, \tau)$, $x_{10}(\xi, \tau)$, and $F(\tau, \zeta)$, which describe the surface of the body, the pressure distribution over the body, and the entropy distribution along the streamlines, respectively. We represent the solution in the region adjacent to the front in the form

$$
\begin{gather*}
x_{i}=\tau^{i-1}\left(y_{i} \div \tau^{2} z_{i} \div \ldots u_{1}=1+\tau^{2} v_{1}+\tau^{4} u_{1} \div \ldots\right. \\
\rho=\rho_{0}-\tau^{2} \rho_{1} \div \ldots u_{2}=\tau\left(v_{2}+\tau^{2} w_{2}+\ldots\right.  \tag{7}\\
p=\tau^{2}\left(\xi+\tau^{2} p_{1}+\ldots \psi=\tau^{1}+v \zeta .\right.
\end{gather*}
$$

The expansions (7) are obtained in accordance with the orders of magnitude of the unknown quantities at the front. The functions $p_{i}$ are given only through their values at the front, for the satisfactorion of which it is sufficient to take $p_{i}=p_{i}(\xi)$. Changing to the new independent variables $\xi$ and $\zeta$, after the substitution of ( 7 ) into (1) we obtain

$$
\begin{align*}
& \frac{\partial v_{i}}{\partial \xi}=(-1)^{i} y_{i}^{v} \frac{\partial y_{i+1}}{\partial G} ; \quad \frac{\hat{\partial}}{\partial_{\xi}^{E}}\left[\frac{\xi}{\rho_{U}^{x}}\right]=0 ; \\
& \frac{\partial u_{i}}{\hat{\sigma}_{\xi}^{*}}-\frac{\partial p_{1}}{\hat{\sigma}_{\xi}} \frac{\partial v_{i}}{\partial \xi}=(-1)^{i}\left[y_{2}^{i} \frac{\partial z_{i+1}}{\partial_{\xi}^{\prime}}+v \frac{\partial y_{i+1}}{\partial \xi} z_{2}\right] ; \quad \frac{\partial}{\partial \xi}\left[\frac{p_{1}}{\xi}-\gamma \frac{\rho_{1}}{\rho_{0}}\right]=0 ; \\
& v_{1}+\frac{1}{2} v_{2}^{2}+\frac{x}{x-1} \frac{\xi}{\rho_{0}}=\frac{1}{x-1} K^{-2} ;  \tag{8}\\
& u_{1}+\frac{1}{2} v_{1}^{2}+v_{2} u_{2}+\frac{1}{2} w_{2}^{2}+\frac{x}{x-1}\left(\frac{p_{1}}{\rho_{0}}-\frac{\xi \rho_{1}}{\rho_{0}^{2}}\right)=0 ; \\
& \frac{\partial y_{1}}{\partial \xi} v_{2}=\frac{\partial y_{2}}{\partial \xi} ; \quad \frac{\hat{\sigma}_{-}}{\partial_{\xi}^{5}} v_{2}+\frac{\hat{\partial} y_{1}}{\partial_{5}^{c}} u_{2}=\frac{\partial z_{2}}{\partial \xi}+v_{1} \frac{\partial y_{2}}{\partial \xi} .
\end{align*}
$$

The expression for the normal to the front has the form

$$
\begin{gathered}
n_{1}=-\tau A\left(1+\tau^{2} A^{2}\right)^{-1 / 2} ; n_{2}=\left(1+\tau^{2} A^{2}\right)^{-1 / 2} \\
A=2^{-v / 2} \zeta-v /(1+v) d g^{-1} / d \zeta
\end{gathered}
$$

and by expanding this in series which converge for $\tau^{2} A^{2}<1$ and substituting the expansions into the boundary conditions at the front, we obtain

$$
\begin{gather*}
\xi=[2 /(x-1)] A^{2}+[(1-x) / x(x+1)] K^{-2} \\
v_{1}=-[2 /(x-1)] A^{2}+\left[2 /(x+1) \mid K^{-2}\right. \\
p_{1}=-[2 /(x+1)] A^{4} ; w_{1}=[2 /(x+1)] A^{4} \\
1 / \rho_{0}=(x-1) /(x+1)-[2 /(x-1)] K^{-2} A^{-2}  \tag{9}\\
v_{2}=\left\{[2 /(x-1)] A^{2}-[2 /(x+1)] K^{-2}\right\} A^{-1} \\
\frac{\rho_{1}}{\rho_{0}^{2}}=-\frac{2}{x+1} K^{-2} ; \quad w_{2}=-\frac{2}{x+1} A^{3}
\end{gather*}
$$

Considering the condition $\tau \rightarrow 0$, we will assume that the boundary conditions (9) are satisfied for $\zeta \geqslant 0$. Thus, the problem becomes closed for each approximation of the outer
expansion (7). The solution of the zeroth approximation describes plane and axisymmetric one-dimensional unsteady gas flows behind a front of the form $t=g(\zeta), r=2^{\nu / 2} \zeta^{1 /(1+\nu)}$, where $t$ is the time; $r$ is the spatial coordinate. It is shown in [7] that for the existence of a solution of the unsteady problem it is necessary that $\alpha \leqslant(3+\nu) / 2(1+v)$. From this follows the limitation imposed above on $\alpha$.

The joining of (3)-(6) only with the zeroth approximation of the expansion (7) is carried out below. Since the zeroth approximation itself is an asymptotic representation of the exact solution outside the inner region, the terms of the zeroth approximation retained in the joining must be of a lower order of smallness than the principal terms of the higher approximations. And the choice of the region of joining is not obvious without an analysis of the behavior of the higher approximations. Therefore, it seems appropriate to make a brief study of the behavior of the higher approximations.

If in a region $R$ adjacent to the front, except for some vicinity of the lines $\zeta=0$, the boundary conditions provide for a sufficiently smooth solution of the zeroth approximation, then the continuous solution of the higher approximations up to some number $j$ is determined in this region. To discover the nature of the behavior of the latter in the vicinity of the line $\zeta=0$ we represent the solution of the $k$-th approximation in the form of the asymptotic expansions

$$
F_{k}=F_{k 0}(\xi) g_{k 0}(\zeta)+F_{k 1}(\xi) g_{k 1}(\xi) \div \ldots, g_{k, j \div 1} 1 / g_{k, j \rightarrow 0} \text { as } \zeta \rightarrow 0 .
$$

Substituting the latter into (8) and allowing for the conditions (9), we obtain

$$
\begin{gathered}
\left.\because \sim \Delta^{-2} \Phi ; u_{1} \sim 1-\tau^{2} \Delta^{2} \Phi ; x_{1} \sim 0(1) \therefore \tau^{2} \Delta^{2}\right\} \Phi ; \\
x_{2} \sim \tau\left(0(1)+\tau^{2} \xi \Delta^{2}+2 x(I) ; u_{2} \sim \tau\left(0(1) \sim \tau^{2} \Delta^{2} \Phi\right) .\right.
\end{gathered}
$$

if $y_{20} \neq 0$. In the case of $y_{20}=0$ the only difference will be with $v=1$ for the components

$$
\begin{gathered}
x_{1} \sim 0(1)+\tau^{2} 0\left(\Delta^{21 / 2}, \int_{0}^{\zeta} \Delta^{2 x} d \zeta\right) \Phi ; \quad x_{2} \sim \tau \Delta \zeta^{1 / 2} \Phi ; \\
u_{2} \sim \tau \Delta \xi^{1} \eta^{2}\left(0(1)+\tau^{2} \Delta^{2 x} \Phi\right) ; \Phi=0(1)+\tau^{2} 0\left(\Delta^{2 x}\right)+\tau^{4} 0\left(\Delta^{1 x}\right) \ldots \ldots
\end{gathered}
$$

 scribes the nature of the behavior of the $j$-th approximation in the vicinity of the line $\zeta=0$ relative to the zeroth approximation. Thus, the rate of decline of the asymptotic expansions with respect to $\tau$ is characterized by the expression $\tau^{2} 0\left(\Delta^{2 k}\right)=\left(\zeta^{2}[1 /(1+\nu)-\alpha]\right)$. Therefore, the corrections to the zeroth approximation are generally not small in the region of $\tau^{2} 0\left(\Delta^{2 k}\right)$ ~ 1 , for which the values of the entropy in the zeroth approximation are close to the values of the entropy behind a straight compression shock in a steady stream. The fact of the necessity of a correction to the zeroth approximation for these values of $\zeta$ is obtained from other considerations in [3].

Being confined to only the zeroth approximation of the outer expansion, we determined the arbitrary functions $x_{20}$ and $x_{10}$ of (5) and (6) from the condition of joining in the region $\tau^{2} 0\left(\Delta^{2} K\right) \sim \tau^{\beta_{0}}$, where the zeroth approximation and (5) have the same accuracy. Using the boundary conditions at the front, one can show that in this case $\beta_{0}=2 /(\kappa+1)$ and one must set $\gamma=\beta_{0}$ in (5) and (6), while the region of joining is determined by the values of $E \sim 0$ $\left(\tau^{\kappa(1+v) /(x+1)(\alpha(1+v)-1)}\right.$. The solution of the zeroth approximation for the $\zeta$ of this region has the form

$$
\begin{gathered}
y_{2}^{1+v}=y_{20}^{1+v}(\xi)+2^{v} \int_{0}^{\zeta} \frac{1}{\rho_{0}} d \zeta\left(1+0\left(\tau^{\beta_{0}}\right)\right) \\
v_{2}=\frac{d y_{10}^{-1}}{d \xi} \frac{\partial}{\partial \xi}\left[y_{20}^{1+v}(\xi)+2^{v} \int_{0}^{\zeta} \frac{1}{\rho_{0}} d \zeta\left(1+0\left(\tau^{\beta_{0}}\right)\right)\right]^{\frac{1}{1+v}}\left(1+0\left(\tau^{\beta_{0}}\right)\right)
\end{gathered}
$$

$$
\begin{aligned}
& c_{1}:=-\frac{\alpha}{x-1} \frac{\xi}{\rho_{0}}+0(1), \quad \frac{1}{\rho_{0}}=c_{v} 5^{-\frac{1}{x}} A^{\frac{2}{x}}, \quad c_{0}=\left(\frac{2}{x+1}\right)^{\frac{1}{x}}\left(\frac{x-1}{x+1}\right) .
\end{aligned}
$$

Here we retain only the terms which are of a lower order of smallness than the principal terms of the higher approximations. The function $t=y_{10}(\xi)$ describes the pressure distribution, while the function $r=y_{20}(\xi)$ describes the trajectory of a particle passing through the front at the start of the motion. The methods of determining the latter are well developed $[1,8]$, and therefore we will henceforth consider the functions $\mathrm{y}_{10}$ and $\mathrm{y}_{20}$ as known. For a detailed analysis of the flow we require the concrete form of the function $g(\zeta)$ in the small vicinity of the apex of the front. We perform the analysis for functions $g(\zeta)$ which for $5 \int_{0}^{0}\left(\mathrm{~T}^{\mathrm{x}}(1+v)(x \div 1)(\alpha: 1+v)-9\right)$ satisfy the condition

$$
\begin{equation*}
A=A_{0}\left(1+0\left(\tau^{\beta_{0}}\right)\right), \quad A_{0}=2^{-v_{0}^{2}}(c x)^{-1} \zeta^{1} /(1 \div v)-x, \tag{10}
\end{equation*}
$$

which clarifies the degree of closeness of the function $g(\zeta)$ to a power-law function. Here $c>0$ is an arbitrary constant. Since $K \geqslant 1$, from the boundary conditions at the front for the indicated values of $\zeta$ we have

$$
\frac{1}{\beta} \cdots c_{0} \xi^{-\frac{1}{x}} \tau \frac{\frac{2}{x}}{x} X\left(1+0\left(\tau^{\beta_{0}}\right)\right), \quad X-\left(\frac{\tau^{2} A_{0}^{\frac{2}{0}}}{1+\tau^{*} A_{0}^{2}}\right)^{\frac{1}{x}} .
$$

Using (10), one can obtain

$$
\begin{equation*}
\int_{0}^{\zeta} \frac{1}{\rho f} d_{s}^{c} c_{0} s^{-\frac{1}{x}} \tau-\frac{2}{x} \int_{i}^{\zeta} X\left(1+\sum_{j=1}^{1} \frac{\eta(\eta-1) \ldots(\eta-j+1)}{j!}\left(B \tau^{2-\frac{2}{x}} X\right)^{j} d \zeta\left(1+0\left(\tau^{\beta_{0}}\right)\right),\right. \tag{11}
\end{equation*}
$$

where $B=-[2 k /(\kappa-1)] c_{0} \xi^{1-1 / k} ; \eta=-1 / 2$; the integer $Z$ is determined below. In the case of $\alpha=(3+v) / 2(1+v)$ (a strong explosion in the corresponding unsteady problem with subsequent expansion of the piston) we have

$$
\begin{equation*}
\int_{0}^{\frac{1}{2}} X^{n} d \xi=\frac{x}{x-n} \mu\left(X^{n-x}-1\right), \mu=c^{-2}\left(\frac{4}{9}\right)^{1-v} 2^{-v} \tau^{2}, \quad n=1,2, \ldots \tag{12}
\end{equation*}
$$

In the general case for $\zeta$ from the region of joining we have

$$
\begin{equation*}
\int_{0}^{\xi} X^{n} d \zeta=\frac{1+v}{2(\alpha(1+v)-1)}\left(\tau F_{0}\right)^{\frac{1+v}{\alpha(1+v)-1}}\left(\Phi_{1}(b)-\Phi_{0}(\xi)\right) . \tag{13}
\end{equation*}
$$

Here and below $\mathrm{F}_{0}=2^{-\nu / 2}(\mathrm{c} \alpha)^{-1}, \mathrm{~b}=-\mathrm{n} / \mathrm{k}, a=-\mathrm{b}-\mathrm{h}-2, \mathrm{~h}=(1+\nu) / 2(\alpha(1+\nu)-1)-1$; the function $\Phi_{0}$ has the form

$$
\begin{aligned}
& 0\left(\tau^{2} A_{0}^{2}\right), a+1 \neq 0, \\
& 0 \ln \left(\tau^{2} A_{0}^{2}\right), \quad a+1=0,
\end{aligned} \quad \text { if } \quad 1 / \%<-b,
$$

We note that when $b=-1 / \kappa$ the number $a+1 \neq 0$ for the intervals of variation of $\alpha$ and $\kappa$ under consideration, while the residual term must be replaced by $0 \ln \left(\tau^{2} A_{0}{ }^{2}\right)$ with $\alpha=1 /(1+$ $v)+\kappa / 2(\kappa+1)$. The constant $\Phi_{1}$ does not depend on $\tau$ and has the form: 1) $-\mathrm{b}<2$,

$$
\Phi_{1}=[1 /(h+1)] F(-b, h \div 1, h \div 2,-1)+[1 /(a \div 1)] F(-b, a+1, a+2,-1)
$$

2) $2 \leqslant-b<3$,

$$
\begin{gathered}
\Phi_{1}=-2^{b+2}(b+1)^{-1}+(h+1)^{-1}(b+1)^{-1}(b+h+2) F(-b-1 \\
h+1, h+2,-1)+(a+1)^{-1}(b+1)^{-1}(b+a+2) F(-b-1, a+1, a+2,-1)
\end{gathered}
$$

3) $-\mathrm{b} \geqslant 3$; $\mathrm{m}=[-\mathrm{b}]-1$ ([ ] is the integral part of the quantity),

$$
\begin{aligned}
& \Phi_{\mathbf{1}}=-\frac{2^{b+2}}{b+1}-\sum_{k=2}^{m} \frac{(b+h+2) \ldots(b+h+h)+(b+a+2) \ldots(b+a+k)}{(b+1) \ldots(b+k)} \times \\
& \times 2^{b+k}+\frac{(b+h+2) \ldots(b+h+m+1)}{(b+1) \ldots(b+m)} \frac{1}{h+1} F(-b-m, h+1, h+2,-1)+ \\
& \quad+\frac{(b+a+2) \ldots(b+a+m+1)}{(b+1) \ldots(b+m)} \frac{1}{a+1} F(-b-m, a+1, a+2,-1),
\end{aligned}
$$

where $F(\alpha, \beta, \gamma,-1)$ are converging hypergeometric series [9]. Possible singular cases are included in 1-3) by the following agreement: if $a+j+1=0, j=0,1, \ldots$, then the $j$-th term in the corresponding series is replaced by zero.

Using (11) and (13), for $\zeta$ from the region of joining we have

$$
\begin{align*}
& \left(\frac{x_{2}}{\tau}\right)^{1+v}-y_{2}^{1+v}=x_{20}^{1+v}(\xi, \tau)-y_{20}^{1+v}(\xi)+\Phi_{2}+0(\varepsilon), \\
\Phi_{2}= & 2^{v-1} c_{0} \xi^{-\frac{1}{x}} \frac{1+v}{\alpha(1+v)-1}\left(F_{0} \tau\right)^{\frac{1+v}{\alpha(1+v)-1}} \tau-\frac{2}{x}  \tag{14}\\
+ & \Phi_{1}\left(-\frac{1}{x}\right)+ \\
& \left.\frac{\eta(\eta-1) \ldots(\eta-j+1)}{j!}\left(B \tau^{2-\frac{2}{x}}\right)^{j} \Phi_{1}\left(-\frac{j+1}{x}\right)\right] .
\end{align*}
$$

The expressions $[\alpha-1 /(1+v)]^{-k} \tau^{(1+v) /[\alpha \alpha(1+v)-1]}$, where the integer $k<2$, which are encountered here behave in a nonsingular way as $\alpha \rightarrow 1 /(1+\nu)$ as the function $k!(\ln \tau)^{-k} \tau^{(1+v)}$ $[\alpha(1+v)-1]$. The residual term in (14) has the order $0\left(\tau^{2} \ln \tau\right)$ when $\alpha=1 /(1+v)+\kappa / 2(\kappa+1)$ and $0\left(\int_{0}^{5} \frac{1}{\rho_{0}} d \zeta \tau^{\beta_{0}}\right) \sim \tau^{\frac{x(1+v)}{(\alpha+1)(\alpha(1+v)-1)}}$ in the joining region in the remaining cases.

From a comparison of the orders of magnitude of the function $\Phi_{2}$ and the residual term it follows that for $1 /(1+v)<\alpha \leqslant 1 /(1+v)+\kappa / 2(\kappa+1)$ in the determination of the function $\mathrm{X}_{2}$ o the function $\Phi_{2}$ must be included in the residual term. As one can show for $1 /(1+$ $v)+\kappa / 2(\kappa+1)<\alpha \leqslant(3+v) / 2(1+v)$ the principal term $\varepsilon_{0}$ of the function $\tau^{2} z_{2}$ of the first approximation in the joining region is of the same order of magnitude as the residual term in (14), while for $1 /(1+\nu) \leqslant \alpha \leqslant 1 /(1+\nu)+\kappa / 2(\kappa+1)$ it does not exceed $0\left(\tau^{2}\right)$. Thus, it is incorrect to make the residual term in (14) concrete without allowance for the solution of the first approximation of the expansion (7). Therefore, setting $\mathrm{x}_{20}(\xi, \tau)=$ $\mathrm{y}_{20}(\xi)$ in the case of $1 /(1+v) \leqslant \alpha \leqslant 1 /(1+v)+\kappa / 2(\kappa+1)$ and setting $\mathrm{x}_{20}{ }^{1+\nu}(\xi, \tau)=\mathrm{y}_{20}{ }^{1+}+\nu(\xi)$ $-\Phi_{2}$ and $l=\left[\left(2-\frac{2}{x}\right)^{-1}\left(\frac{2}{x}-\frac{1+v}{(x+1)(\alpha(1+v)-1)}\right)\right] \quad$ in the case of $l /(1+\nu)+\kappa / 2(\kappa+1)<\alpha /$ $(3+v) / 2(1+v)$ (in this case for $Z=0$ one must keep only the first term in the function $\Phi_{2}$ ), we obtain in the joining region $\left(\mathrm{x}_{2} / \tau\right)^{1+\nu}=\mathrm{y}_{2}{ }^{1+\nu}+0\left(\varepsilon, \varepsilon_{0}\right)$. In the case of $\alpha=$ (3+ $v) / 2(1+v)$ the expression for $\Phi_{2}$ is simplified in accordance with (12) and has the form

$$
\begin{equation*}
\Phi_{2}=-2^{\nu} c_{0} \xi^{-\frac{1}{x}} \tau-\frac{2}{x} \mu\left(\frac{x}{x-1}+\sum_{j=1}^{l} \frac{\eta(\eta-1) \ldots(\eta-j+1)}{j!}\left(B \tau^{2-\frac{2}{x}}\right)^{j} \frac{x}{x-j-1} .\right) \tag{15}
\end{equation*}
$$

The functions $x_{1}$ and $y_{1}$ are joined in accordance with the behavior of the function $y_{20}$. Taking $x_{10}=y_{10}(\xi)$ in (6), we obtain the following:
A. $y_{20} \neq 0$. In the joining region

$$
x_{1}=y_{10}(\xi)+\overline{y_{20}} \frac{d}{d \xi}\left(\frac{d}{d \xi}\left(y_{20}\right) \frac{d y_{10}^{-1}}{d \xi}\right) \zeta+0\left(\varepsilon_{1}\right) \zeta, \quad x_{1}-y_{1}=0\left(\varepsilon_{1}\right) \zeta .
$$

The quantity $0\left(\varepsilon_{2}\right)$ does not exceed in order of magnitude the maximum of $\tau^{\beta_{0}}, \tau^{2-2 / \kappa}$, $\tau^{\frac{1 x(1+v)}{(x+1)(\alpha(1+v)-1)}-\frac{2}{x+1}}$. Thus, with an accuracy $0\left(\varepsilon_{1}\right) \frac{x(1+v)}{(x+1)(\alpha(1+v)-1)}$ in comparison with unity $\mathrm{x}_{1}=\mathrm{y}_{1}$ in the joining region.
B. $y_{20}=0$. 1) $v=0$. In the joining region

$$
x_{1}=y_{10}(\xi) \div 0\left(\tau^{\frac{1+v}{x(1+v)-1}-\frac{2}{x}}\right) \div 0\left(\int_{0}^{\infty} \frac{1}{\rho} d \xi d \xi\right),
$$

and, as one can show,

$$
x_{1}-y_{1}=\left(1 \frac{2 x(1+v)}{\tau^{(x \cdot 1)(\alpha(1-v)-1)}-\frac{2}{x-1}}: \quad \tau^{\frac{x(1+v)}{(x+1)(\alpha(1+v)-1)}+\frac{2}{x+1}}\right)
$$

2) $v=1$. In the joining region

$$
\begin{gathered}
x_{1}: y_{10}(\xi)+\xi^{\frac{1}{2 x}} \frac{d}{d \xi}\left[\frac{d}{d \xi}\left(\xi^{-\frac{1}{2 x}}\right) \frac{d y_{10}^{-1}}{d_{5}^{c}}\right] 5+0\left(\tau^{2-\frac{2}{x}}, \tau^{\beta_{0}}\right) \zeta \\
x_{1}-y_{1}=0\left(\varepsilon_{1}\right) \tau^{\chi(1+w)(x+1)\left(\alpha_{0}(1+v)-1\right)}
\end{gathered}
$$

After the arbitrary functions $x_{20}$ and $x_{10}$ are determined, one can verify that the joining of the remaining functions is performed with an accuracy $0\left[\tau^{2 /(\kappa+1)}\right.$ in comparison with unity, while a solution which is uniformly valid in the region under consideration is constructed as follows. In the outer region the solution is described by the zeroth approximation of the expansion (7), while in the inner region it is described by the functions

$$
\begin{aligned}
& \left(\frac{x_{2}}{\tau}\right)^{1 / v} \quad x_{21}^{1+v}(\xi, \tau)+2^{x} \int_{0}^{\frac{1}{1}} \frac{1}{\rho f} d \xi, \quad u_{1}=f, \\
& x_{1}=y_{10}(\xi) \therefore x_{11} \xi, \quad \frac{u_{2}}{\tau} \frac{\partial x_{1}}{\alpha_{\xi}}=f \frac{\partial}{\partial \xi}\left(\frac{x_{2}}{\tau}\right), \quad 1 / 0=c_{0} \xi^{-1 / x} \tau^{-2 / x} X
\end{aligned}
$$

(the functions $x_{20}$ and $x_{1,1}$ were determined above).
The calculation of the integral is simplified by the fact that the pressure enters into the integrand as a parameter, and it is actually performed during the determination of the arbitrary function $\mathrm{x}_{20}$. Here the outer and inner regions, as shown above, have a common part where both expansions are applicable. The shape of the surface $\zeta=0$ of the body has the form

$$
\left(\frac{x_{2}}{\tau}\right)^{1 \div v}=y_{20}^{1+v}(\xi)+\left\{\begin{array}{l}
0\left(\begin{array}{c}
\frac{x(1+v)}{\tau^{(\alpha-1)(x(1+v)-1)}}, \varepsilon_{0}
\end{array}\right), \quad \frac{1}{1+v}<\alpha<\frac{1}{1+v}+\frac{x}{2(x+1)} \\
-\left(\mathrm{J}_{2}, \quad \frac{1}{1-v}+\frac{x}{2(x-1)}<\alpha \leqslant \frac{3-v}{2(1+v)}\right. \\
x_{1}=y_{10}(\xi) .
\end{array}\right.
$$

Thus, if one considers only the zeroth approximation of the expansion (7), then as $\alpha \rightarrow$ $1 /(1+v)$ the correction to the contour of the body obtained from the solution of the corresponding unsteady problembecomes as small as desired in the region under consideration. The transverse size of the region of inapplicability of the zeroth approximation of the expansion (7), determined above by the values of $\zeta \leqslant 0\left[\tau^{\kappa}(1+v) /(\kappa+1)(\alpha(1+v)-1)\right]$, also becomes as small as desired, which reflects the fact of the continuous dependence of the solution on the boundary conditions, since when $\alpha=1 /(1+v)$ the entire field of flow is described by the zeroth approximation [1]. However, for all $1 /(1+v)<\alpha \leqslant(3+v) / 2(1+v)$ the correction to the zeroth approximation of the expansion (7) in the inner region is not small.

In comparing particular cases of the solution obtained with known solutions we note that if the shape of the front in the vicinity of the apex has the form $x_{2} / \tau=\kappa_{1} x_{1}{ }^{2} /(3+v)$, then the results of [10] are suitable for the determination of the shape of the surface of the body. And, as follows from [10], the equation of the surface of the body has the form

$$
\left(\frac{x_{2}}{\tau}\right)^{1+v}=Y_{b 0}^{1+v}\left(x_{1}\right)+p_{00}^{-\frac{1}{x}}\left(x_{1}\right) \frac{x}{x+1}\left(\frac{2}{x+1}\right)^{\frac{1}{x}}\left(\frac{2}{3+v}\right)^{2} x_{1}^{3+v} \tau\left(\frac{\frac{x}{x}-1}{x}\right),
$$

where $\mathrm{Y}_{\mathrm{b}}$ is the shape of the body and poo is the pressure distribution over the body, which follow from the solution of the corresponding unsteady problem. In this case, in [10] terms of order $0\left(\tau^{4}\left[\left(\kappa^{-1}\right) / k\right]\right.$ ) are discarded in the calculation. One can verify that this case corresponds to the following values of the parameters used above: $\left.\kappa_{1}=2^{v / 2} c^{-1 / \alpha(1}+\nu\right), \alpha=$ $(3+\nu) / 2(1+\nu)$. Retaining in (15) only the term of order $0\left(\tau^{2}\left(k^{-1}\right) / k\right)$ and using the expression obtained above for the shape of the body, we obtain

$$
\left(\frac{x_{2}}{\tau}\right)^{1+v}=y_{20}^{1+v}(\xi)+\frac{x}{x+1}\left(\frac{2}{x+1}\right)^{\frac{1}{x}}\left(\frac{4}{9}\right)^{1-v} c^{-2 \xi} \xi^{-\frac{1}{x}} \tau^{2-\frac{2}{x}}, \quad x_{1}=y_{10}(\xi) .
$$

Considering the meaning of the functions $y_{20}$ and $y_{10}$ and the connection $c=c\left(k_{1}\right)$, we see that the two expressions fully coincide. Here the terms of higher order of smallness than the first term in (15) are due to the allowance for the departure of the function $f$ from unity. As follows from (15), the allowance for the latter makes sense for values of $k$ near unity.

In the case when the shape of the front everywhere has the form $\mathrm{X}_{2}=\mathrm{c}_{1} \mathrm{x}_{1}{ }^{\mathrm{n}}$ and $\mathrm{K}=\infty$ (the self-similar case in the corresponding unsteady problem), the solution as $x_{1} \rightarrow \infty$ is constructed in [3]. Let us make a comparison, taking $c_{1}=2^{\nu / 2} c^{-n_{\tau}}, n=1 / \alpha(1+\nu)$, and being limited for simplicity to values of $1 /(1+v)+\kappa / 2(k+1)<\alpha<(3+v) / 2(1+v)$ and to the region of $\zeta \leqslant\left(\tau F_{0}\right)\left({ }^{1}+v\right) /\left[\alpha\left({ }^{1}+v\right)-^{1}\right]$ (for $\zeta$ from this region the projection of the vector of the unit normal to the front onto the $x_{1}$ axis is equivalent to unity in order of magnitude). The departure of $u_{1}$, the longitudinal component of the velocity of the disturbed stream, from unity was neglected in the determination of the coordinate $x_{2}$ in [3]. In the present report the departure was taken into account by the function $f$ and entered into the function $\Phi_{2}$ in the form of the terms additional to the first term. It is therefore appropriate to make the comparison by taking $f=1$ and leaving only the first term in the expression for $\Phi_{2}$.

From the results presented above we get the form for the function $\mathrm{x}_{2}$ in the indicated region:

$$
\begin{gathered}
\frac{x_{2}}{\tau}=y_{\mathrm{b}}\left(x_{1}\right)-\xi_{\mathrm{b}}^{-\frac{1}{x}} y_{\mathrm{b}}^{-v} c_{0_{1}} \tau-\frac{2}{x}\left(A_{1}-\xi F\left(\frac{1}{x}, \frac{k}{2}, \frac{k}{2}+1,-\tau^{-2} F_{9}^{2} \zeta^{\frac{2}{k}}\right)\right), \\
A_{1}=\left(F_{0} \tau\right)^{h}(F(1 / x, k / 2, k / 2+1,-1)+[\chi k /(2-\alpha k) \mid F(1 / x, 1 / x-k / 2,1 / x-k / 2+1,-1)) ;
\end{gathered}
$$

where $k=[\alpha-1 /(1+v)]^{-1} ; \quad y_{\mathrm{b}}=2^{v / 2} c^{-1 / \alpha(1+v)} \lambda_{\lambda_{0}} x_{\mathrm{l}}^{1 / \alpha(1+v)} \quad$ is the surface of the body and $\xi_{\mathrm{b}}=$ $\frac{2^{1-v}}{x+1} \alpha^{-2} c^{-\frac{2}{\alpha(1+v)}} h_{0} x_{1}-2\left(1-\frac{1}{\alpha(1+v)}\right)$ is the pressure distribution over the body which follows from the solution of the unsteady problem; $\lambda_{0}$ and $h_{0}$ are constants. The corresponding expression from [3], rewritten in the notation of the present report, has the form

$$
\frac{x_{2}}{\tau}=y_{\mathrm{b}}\left(x_{1}\right)-\xi_{-}^{-\frac{1}{x}} y_{\mathrm{b}}^{-v} c_{0} \tau-\frac{2}{x}\left(\left(F_{0} \tau\right)^{h} N-\S F\left(\frac{1}{x}, \frac{k}{2}, \frac{k}{2}+1,-\tau^{-2} F_{0}^{-2} \zeta^{\frac{2}{k}}\right)\right) .
$$

Here N is the product of the gamma functions

$$
N=\Gamma(k / 2+1) \Gamma(1 / z-k / 2) \Gamma^{-1}(1 / x) .
$$

Using the well-known properties of hypergeometric series, one can show that $A_{1}=\left(F_{0} \tau^{k}\right) N$. Therefore, the expressions for $\mathrm{x}_{2}$ fully coincide.

The pressure distribution which follows from the transformation of the function $\mathrm{x}_{1}=$ $x_{1}(\xi, \zeta)$ has the form

$$
p=\tau^{2} \frac{2^{1-v}}{x+1} \alpha^{-2} c^{-\frac{2}{\alpha(1+v)}} x_{1} \frac{2}{h \alpha}\left(h_{0}+\frac{\alpha+1}{2 k} \lambda_{0}^{2-v} c^{\frac{1}{\alpha}} x_{1}^{-\frac{1}{\alpha}} \zeta\right) .
$$

The latter coincides with the corresponding expression of [3] with the accuracy of terms containing the functions $P_{2}$ and $P_{4}$ (notation of [3]). The function $P_{2} \sim 0\left(\tau^{2}\right)$ is determined, as follows from [3], by the solution of the first approximation of the expansion (7). And the
allowance for the term $\mathrm{P}_{4}$ makes a contribution to the pressure distribution no larger than $0\left(\tau^{4-2 / K} \zeta\right)$. In the present report neither of these corrections was taken into account in the derivation of the expression for $x_{1}$. The form of the remaining gasdynamic quantities $u_{1}$ and $u_{2}$ is fully determined by the distributions of the functions $x_{2}, p$, and $\rho$, and therefore, no comparison is made for them.

In the general case the agreement established above allows one to effectively construct a solution to the inverse steady problem any time that a solution is constructed in the corresponding unsteady one-dimensional problem.

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## SEMIEMPIRICAL THEORY OF THE GENERATION OF DISCRETE TONES BY A SUPERSONIC

UNDEREXPANDED JET FLOWING OVER AN OBSTACLE
V. N. Glaznev and V. S. Demin

UDC 533.534 .115
§1. The phenomenon of the generation of a strong discrete tone by a supersonic underexpanded jet flowing over an obstacle was first discovered by Hartmann [1]. There are presently a considerable number of reports devoted to the experimental study of the Hartmann effect [2-4]. However, the mechanism of formation of these oscillations has not been clarified up to now [2, 5]. An elementary theory of this phenomenon is presented in the present report.

A diagram of a supersonic underexpanded jet flowing over a flat obstacle is presented in Fig. 1 ( 1 is the jet boundary; 2 is the central compression shock (the Mach disk); 3 is the suspended shock; 4 is the reflected shock; and 5 is the contact discontinuity). The effect consists in the fact that the flow becomes unstable at certain values of the nozzle Mach number $M_{a}$, degree of nonratedness $n=p_{a} / P_{s}$ of the jet ( $P_{a}$ is the pressure in the jet

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